Clean Sky 2 Information Day dedicated to the 10th Call for Proposal (CfP10)

AIFRAME ITD Topics related to AIR WP B.4-2 – Next Generation Civil TiltRotor Project

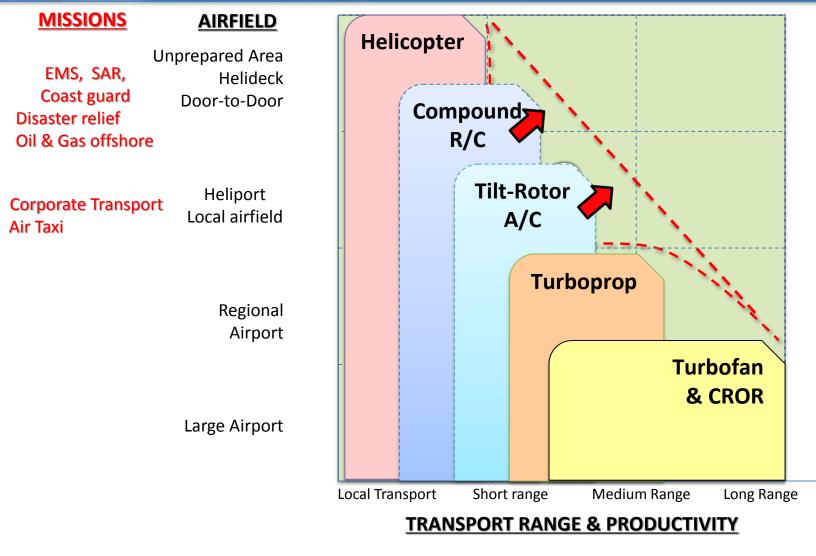
Leonardo Helicopters

Innovation Takes Off

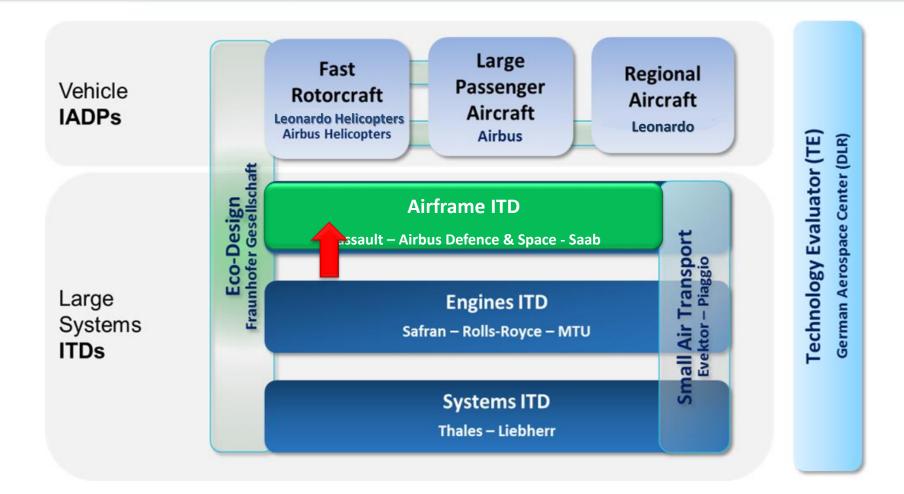
www.cleansky.eu



FRC Overview Filling the Mobility Gap

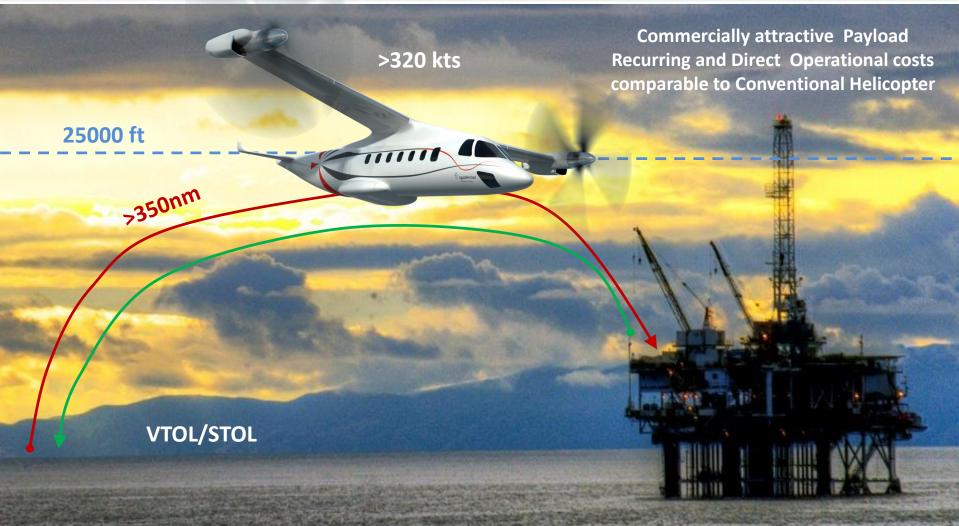


Airframe ITD Overview Clean Sky 2 Context





FRC Overview NextGenCTR Objectives and Challenges



Declared CleanSky 2 Objectives: Low environmental impact with high productivity and efficiency

Approach for NextGenCTR Program: 2 Phases

NextGenCTR A 2-Phase Program

Phase 1

Design, Build, Fly a Technology Demonstrator under CS2 NextGenCTR TD

- De-risk program, expand current TR capability
- Prove Architecture, Technologies, Operations
- Supported by external funding
- Develop collaborations and partnerships
- Sow the seeds for future technologies & products
- Technology exploitation and dissemination



if the market is ripe on-going business case development

Phase 2

Develop & Certify a Product *NextGenCTR*

- Tailored for diverse missions
- State-of-the-art Technologies embedded
- Competitive RC & DOCs vs. Helicopters
- No Legacy with AW609 technologies



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Key objectives will be pursued within CS2 by a Technology Demonstrator focusing on the Design & Development effort of Key Enabling Technologies:

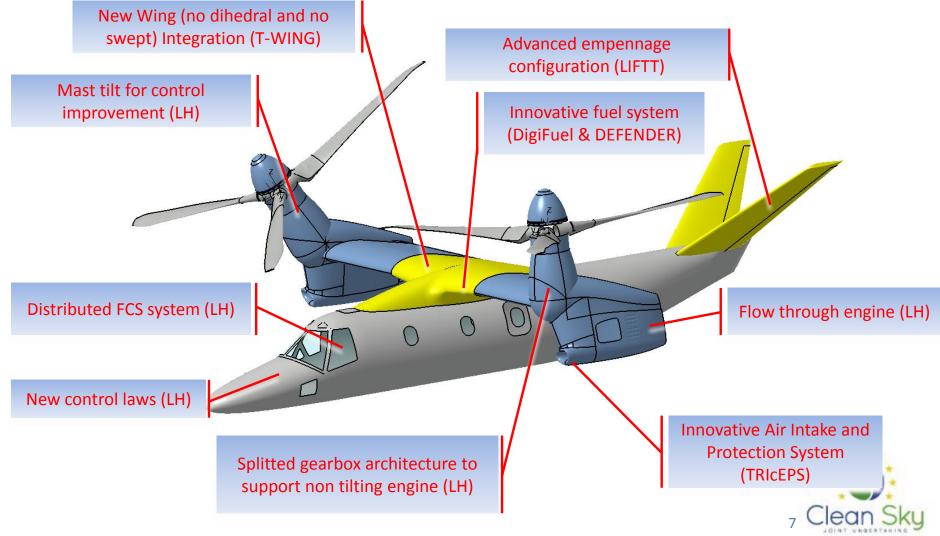
- 1. Fixed-engine, Split Gearbox Drivetrain concept
- 2. Efficient Nacelle architecture
- 3. Advanced Wing architecture
- 4. Optimized Tail configuration
- 5. Advanced Modular, Distributed & Scalable Flight Control System

Throughout this effort the development and validation of predictive models and tools for air vehicle performance (including environmental), efficiency and productivity



NGCTR-TD Objectives

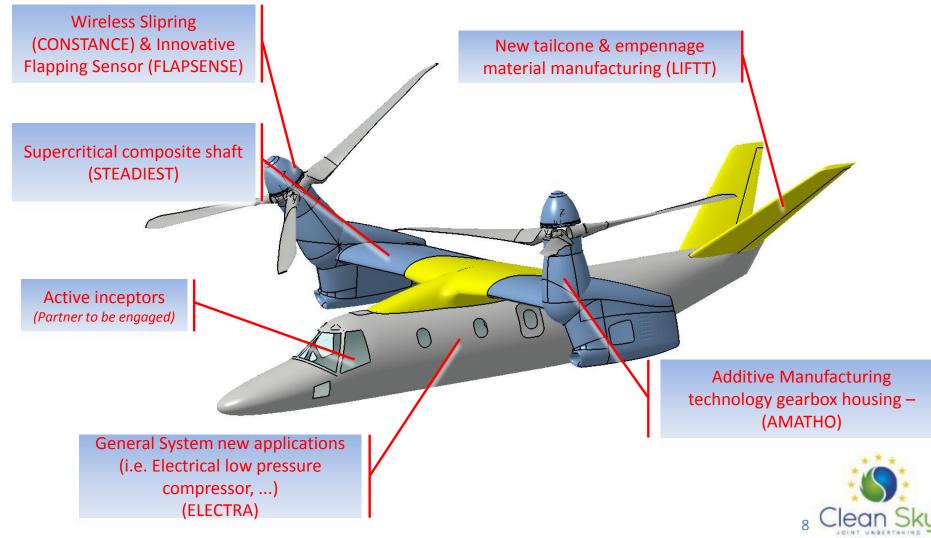
New Technologies to be tested within CS2 for First Flight (TRL =6)



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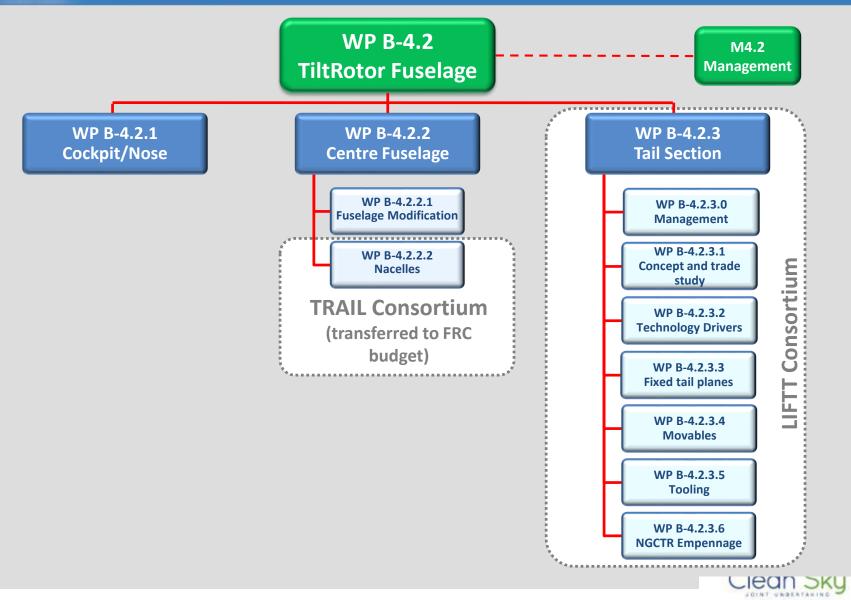
NGCTR-TD Objectives

New Technologies to be tested within CS2 after First Flight (TRL =<6)

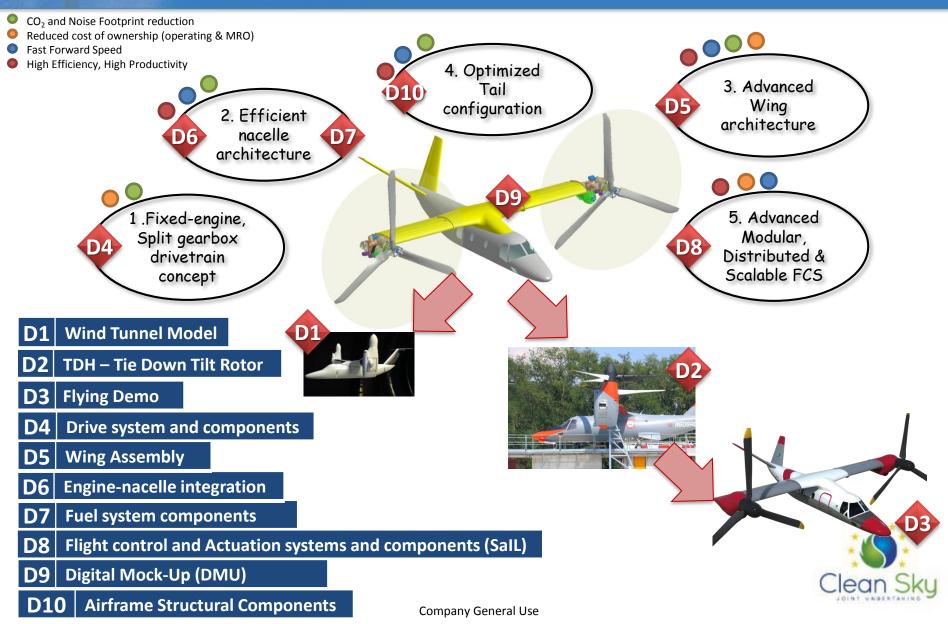


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AIR ITD NGCTR-TD Work Package Structure

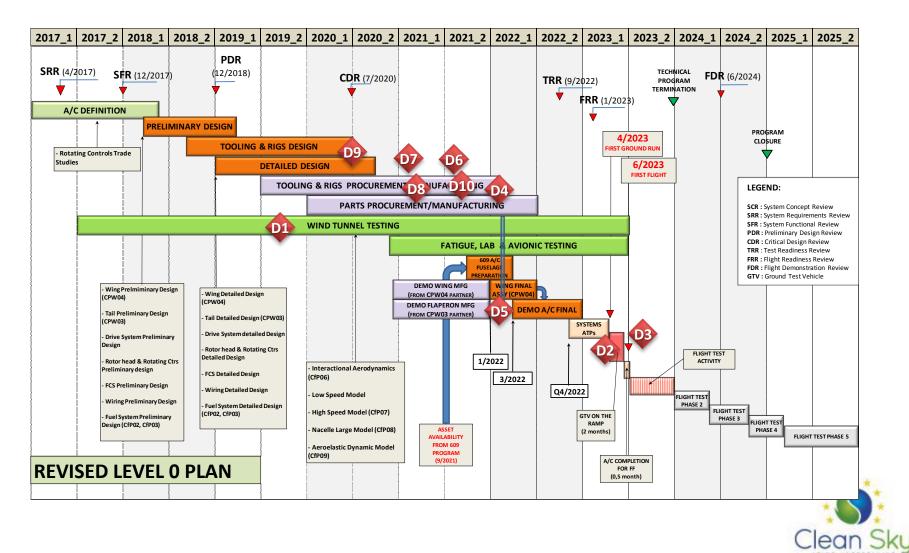


NGCTR WBS - Clean Sky 2 Demonstrators Definition & Plan



NGCTR WBS - Clean Sky 2 Demonstrators Definition & Plan

Master Level "0" Plan Status:



10th Call for Proposal (CfP10) – Fast RotorCraft IADP

- **AIR-02-81:** Active Flow control on Tilt Rotor lifting surfaces
- AIR-02-82: Innovative approaches to interior Noise Control for Next Generation Civil Tilt Rotor
- **AIR-02-83:** Innovative weight measurement system for Tilt Rotor application
- **AIR-02-84:** Modular platform development for Tilt Rotor final assembly

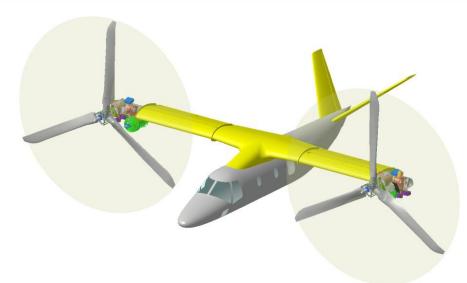


Fast RotorCraft IADP 10th Call for Proposal Open Day

AIR-02-81: Active Flow control on Tilt Rotor lifting surfaces



- Topic Manager: Leonardo Helicopters ;
- <u>Collab.=</u>Implementation Agreement
- Indicative Funding Value: 600 k€;
- Duration: 24 Months
- <u>Type of Action</u>: IA
- <u>Overview</u>: <u>Objectives</u>:



- For the Next Generation Civil Tilt Rotor configuration active devices to maximize the efficiency of the aerodynamic surfaces shall be investigated.
- <u>Objective</u>:

To investigate the application of pulsed air blowing devices with Zero Net Mass Flux (ZNMF) to control the vortical flow on the relevant lifting non-rotating surfaces (empennages, wing) on a Tilt Rotor configuration



Workpackages (cont'd)

WP1: Capture of the relevant flow behaviour

With the support of the Topic Manager, the Applicant shall analyse by CFD the flow field development past the tilt rotor configuration to identify the key and relevant locations prone to separated and vortical flows that can benefit from active devices to control/delay/cancel them.

WP2: Modelling of the ZNMF effects

The Applicant shall model the ZNMF (Zero Net Mass Flux) devices effects by inserting appropriate boundary conditions into the basic flow solver used in WP1. The Applicant shall demonstrates, in the Proposal, to have yet developed and applied a robust methodology to synthetize the complex active devices flow field behaviour when embedded in an external shear flow (It is out of the scope of this project to develop this synthetized method).



Workpackages

WP3: Optimization of the location and performance of active control devices

The Applicant shall now optimize the location and the active parameters of the ZNMF devices, in order to achieve some target aerodynamics benefits, as quantified by the Topic Manager

WP4: Summary and recommendation for ZNMF application

The activity culminates in a summary Report where all benefits/drawbacks of the application of the ZNMF on Tilt rotor surfaces will be described, giving recommendation and guidelines for the installation of such devices to maximise the aerodynamics efficiencies of the surfaces.



Expected capabilities from the Applicant

- Computational resources (hardware and software) suitable for the scopes of the activities in the specified timescale
- Proven capability and skill in flow analysis and optimization
- Proven capability to handle and model active flow controls devices
- Proven capability to manage projects by gathering several and different specialized skills (numerical simulation, flow field analysis, optimization) and demonstrated capability to guarantee the project scheduling and milestones.



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AIR-02-82: Innovative approaches to interior Noise Control for Next Generation Civil Tilt Rotor



- <u>Topic Manager:</u> Leonardo Helicopters ;
- <u>Collab.=</u>Implementation Agreement
- <u>Indicative Funding Value</u>: 600 k€;
- Duration: **36 Months**
- <u>Type of Action</u>: **RIA**
- <u>Overview</u>:

For the Next Generation Civil Tilt Rotor configuration a more accurate procedure to investigate the noise paths and a dedicated active noise control are required.

• <u>Objective</u>:

To investigate innovative approaches for interior Noise Control for Next Generation Civil Tilt Rotor

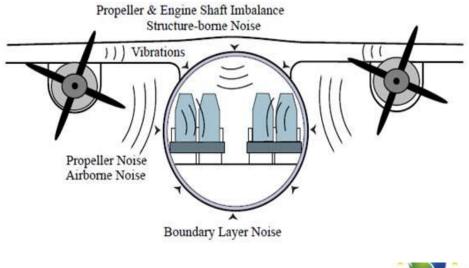


WP1: Advanced TPA

- The Applicant shall quantify and rank the transmission paths of the noise, determining the relative contribution.
- > In order to reach the final target, the applicant shall:
 - Study the partition of the aircraft in several subsystems and choose the significant subsystems of the problem;
 - Analyse the limitations at experimental level for the ATPA test;

Perform the ATPA test on the aircraft. It includes test on ground and at the defined flight conditions.

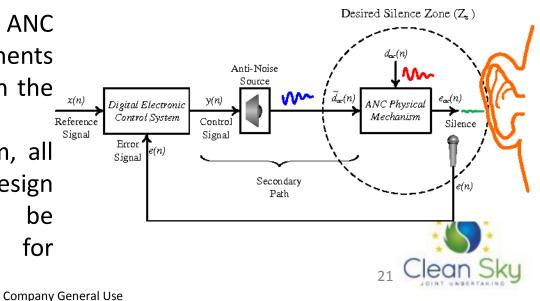
➤ Calculate the numerical parameters from ATPA results (i.e. coupling loss factors, internal loss factor)





WP2: Advanced Active Noise Control (ANC)

- The Applicant shall collect all experimental data (noise and vibration) useful to give the aircraft environment scenario needed for the design phase.
- The Applicant will be responsible for the required inputs (kind and position of the sensors, acquisition system parameters and configuration, etc.), data acquisition and related post-processing.
- The Applicant shall define the ANC system and estimate the expected noise reduction, evaluated using suitable testing and/or mathematical model.
- The Applicant shall define the ANC architecture and all the components will be prepared and installed on the aircraft for testing.
- In order to test the ANC system, all the components defined in the design and development phase will be integrated into the aircraft for performance assessment.



Expected capabilities from the Applicant

- Computational resources (hardware and software) suitable for the scopes of the activities in the specified timescale
- > Proven capability and skill in dynamical testing (FRF, TPA, ATPA) of structures.
- Proven capability to interface experimental and numerical activities (extrapolation of model parameters).
- > Proven capability in the development of ANC architectures for aircraft.
- Proven capability to manage projects by gathering several and different specialized skills (numeric, testing) and demonstrated capability to guarantee the project scheduling and milestones.



Fast RotorCraft IADP 10th Call for Proposal Open Day

AIR-02-83: Innovative weight measurement system for Tilt Rotor application

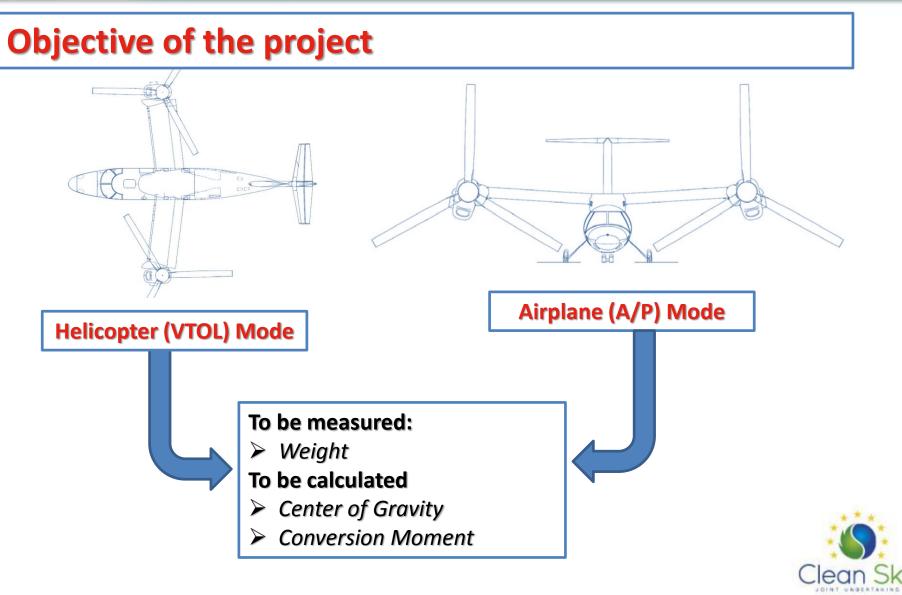


AIR-02-XX: Innovative weight measurement system for Tilt Rotor application

- <u>Topic Manager:</u> Leonardo Helicopters ;
- <u>Collab.=</u>Implementation Agreement
- <u>Indicative Funding Value</u>: **0,8 M€ ;**
- Duration: 24 Months
- <u>Type of Action</u>: IA
- <u>Overview</u>: Measuring and tracking aircraft weight and balance in respect of the technical configuration required for the specific demonstration objective is a key ground testing activity required for release to flight.
- <u>Objectives</u>: The scope of this project is to develop an innovative automatic weight measurement system for tiltrotors aimed at improving the accuracy of weight and balance data (including conversion moment calculations), reducing Flight Line times and improving operators ergonomics



AIR-02-XX: Innovative weight measurement system for Tilt Rotor application



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AIR-02-XX: Innovative weight measurement system for Tilt Rotor application

Main weight management phases to be implemented

Aircraft moved by operator/agv to the station for the Ρ measurement activity R Aircraft is constrained in reference points position 0 Operator select the product program, starts the С automatic measurement routine Ε System lift the footboards in order to level Aircraft to S roll/pitch zero degrees. S Routine is started to detect center of gravity and weight F System returns Aircraft on ground L System record all the data and return a feedback on the 0 conformity of the operation W > Aircraft will return available for next operations

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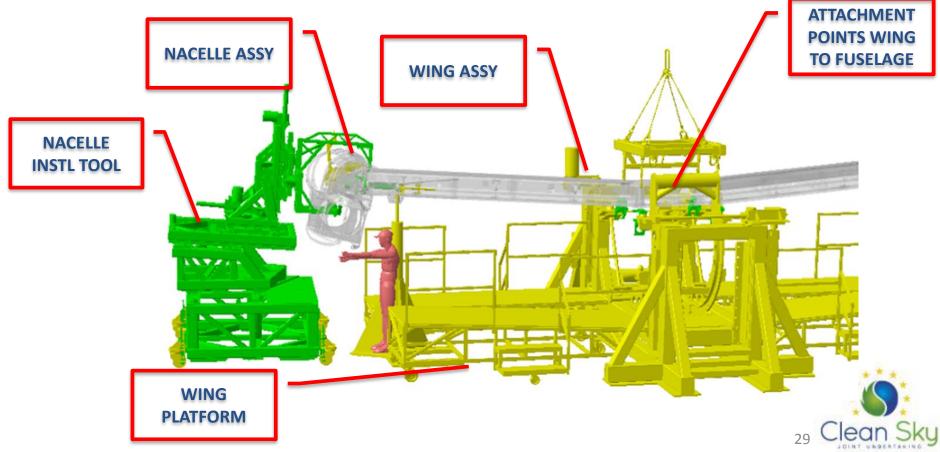
AIR-02-84: Modular platform development for Tilt Rotor final assembly



- <u>Topic Manager:</u> Leonardo Helicopters ;
- <u>Collab.=</u>Implementation Agreement
- <u>Indicative Funding Value</u>: **1,0 M€**;
- Duration: 20 Months
- <u>Type of Action</u>: IA
- <u>Overview</u>: The innovative architecture of the NextGen Civil TiltRotor leads to the need of new suitable methods and tools to ensure the assembly of this demonstrator in a more efficient way, with particular attention to wing assembly, system and subsystems installation and nacelle integration.
- <u>Objectives</u>: The aim of the activity is to design and produce an innovative multifunctional Jig in order to perform structural assembly of NextGen Civil Tilt Rotor wing, wing systems installation, nacelle installation and relevant functional Test



NGCTR ASSEMBLY FLOW- MAIN JIG REQUIRED



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NGCTR ASSEMBLY FLOW- MAIN JIG REQUIRED

WING

ACTIVITY TO BE PERFORMED:

- STRUCTURAL PROVISION INSTL
- NACELLE INSTL
- WING TO FUSELAGE INSTL
- SYSTEMS INSTL

MAIN TARGET TO BE CHECKED:

- > DRIVE SYSTEM SHAFT ALIGNMENT
- > NACELLE ALIGNMENT
- *GOLDEN POINT ALIGNMENT (tbc)*

PLATFORM/JIG/TOOLING NEEDED:

- 1. WING PLATFORM
- 2. WING TO FUSELAGE TOOL
- 3. NACELLE ALIGNMENT TOOL



NGCTR ASSEMBLY FLOW- MAIN JIG REQUIRED

NACELLE

REFERENCE ATTACHMENT POINTS: *TO BE AGREED WITH ENGINEERING*

ACTIVITY TO BE PERFORMED:

> NACELLE BUILD UP:

- > STRUCTURAL PROVISION INSTL
- *SYSTEMS INSTL (PIPES , CABLE, ENGINE, POWERPLANT)*

MAIN TARGET TO BE CHECKED:

> NACELLE ENGINE ATTACHMENT POSITION (TBC)

PLATFORM/JIG/TOOLING NEEDED:

1. NACELLE BUILD-UP STAND



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